



Hochschule für Angewandte Wissenschaften Hamburg

Hamburg University of Applied Sciences

AERO - AIRCRAFT DESIGN AND SYSTEMS GROUP

PreSTo - Aircraft Preliminary Sizing Tool

From Requirements to the Three-View-Drawing

Dieter Scholz Hamburg University of Applied Sciences

EWADE 2011 - 10th European Workshop on Aircraft Design Education Naples, Italy, 24.-27.05.2011



PreSTo:

Aircraft
<u>Pre</u>liminary
<u>S</u>izing
<u>To</u>ol



Contents

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 - PrADO
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Goals

- Give full computer support for the Aircraft Design lecture by Prof. Scholz / Hamburg
- Start tool with nothing but requirements
- Never ask the user for data without giving proper support
- Provide straight forward and fast solutions (=> PreSTo)
- Give the best support (didactics, methods, statistics database, ...)
- Keep user in the loop
- Include expert knowledge in simple "if-then" checks and provide answers with red / green buttons
- Provide aircraft data for 3D-plots and three-view-drawings
- Couple to higher order tools for further investigation

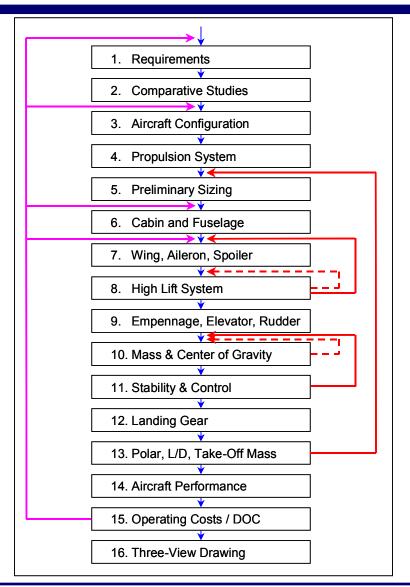




Aircraft Design Lecture

General remarks

- Lecture is based on methods from:
 - Loftin, Torenbeek, Roskam, Raymer, ...
 - Datcom, ...
 - many own additions
- 16 design steps (see Fig.)
- Emphasis on preliminary sizing with matching chart:
 - Jet: T/W = f (m/S)
 - Prop: P/W = f (m/S)
- Lecture in this format since 1999:
 - about 1000 students taught
 - many student reports and theses produced
- Spreadsheet for preliminary sizing (only) is in service for many years: http://FE.ProfScholz.de
- Preliminary sizing spreadsheet has been used for:
 - tutorials, examinations
 - projects, theses

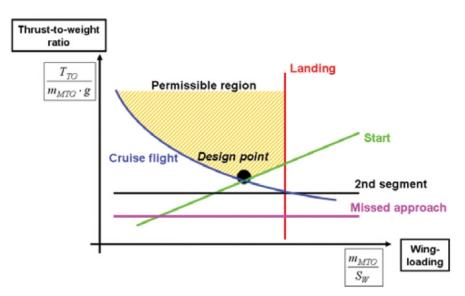


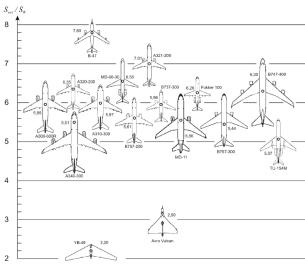




Aircraft Design Lecture Contents

- Preliminary sizing
 - Matching chart
 - (L/D)_{max} estimation with "wetted aspect ratio"
 - Fuel calculation with fuel fractions
- Cabin & fuselage
 - Seats abreast optimum
 - Baggage and cargo volume check
 - Cross section optimization
 - Cabin surface estimation
 - Ditching check: waterline & door sill
 - Exit type and location: check
- Wing
 - Wing parameters found for best operational characteristics
- High Lift
 - High lift geometry found from trial & error procedure
 - C_L_{max} found from Datcom
- Empennage I
 - Sizing from tail volume







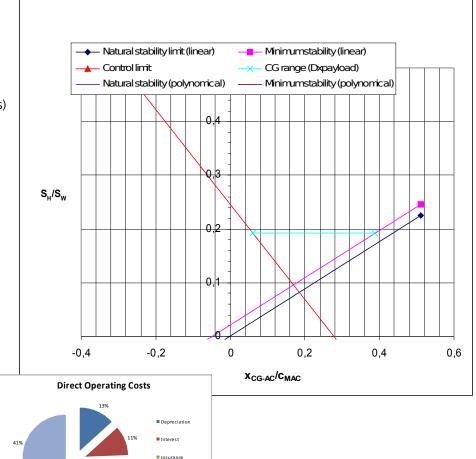


Aircraft Design Lecture - Contents

- Mass and CG
 - Mass from three methods
 - Roskam (OEW distributed about A/C main components)
 - "Modified Raymer" (mass from one key parameter)
 - Torenbeek (well proven)
 - CG determination and wing position correction
 - Loading diagramm (mass versus CG position) for all sensible load cases established
- Empennage II (stability & control power)
 - Horizontal tail
 - Vertical tail
- Landing gear (parameters selected)
 - tip over stability
 - clearance (engine, tail, L/G retraction)
 - Flotation with COMFAA.exe
- Drag
 - Drag from two methods:
 - · wetted area
 - skin friction drag, pressure drag wave drag, interference drag
- Design evaluation:

Direct Operating Coast, DOC

Method: Asssociation of European Airline



■ Maintenance



Screen Shots

PreSTo Control Center and Database



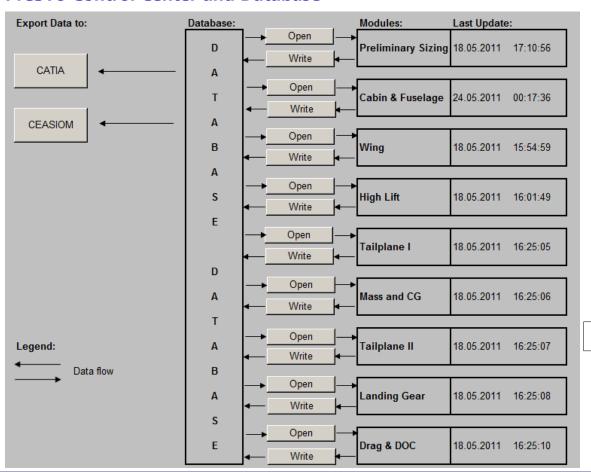
PreSTo Control Center – **Start page**





Screen Shots

PreSTo Control Center and Database



PreSTo Control Center - Module page





Screen Shots

PreSTo Control Center and Database



PreSTo - Aircraft Preliminary Sizing Tool



Version 1.0

PreSTo is an Excel spreadsheets based on Prof. Dieter Scholz' aircraft design lecture. This tool allows the user to quickly design an aircraft and optimise it, starting from the basic requirements such as number of passengers, range or cargo mass to continue with its main parts: fuselage, wing, tail, landing gear,...Besides, masses and position of CG also Direct Operating Costs (DOC) are calculated.

Further analysis in the area of e.g. flight dynamics or CFD is enabled with the connection to CEASIOM. PreSTo further connects to PrADO and CATIA.

For further information, documentation and downloads see:

http://PreSTo.ProfScholz.de

PreSTo is a project by:

Aero - Aircraft Design and Systems Group
Department for Automotive and Aeronautical Engineering
Hamburg University of Applied Sciences (HAW Hamburg).

http://Aero.ProfScholz.de http://www.fzt.haw-hamburg.de http://www.haw-hamburg.de





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PreSTo is distributed in the hope that it will be useful, but WITHOUT ANY WARRANTY; without even the implied warranty of MERCHANTABILITY or FITNESS FOR A PARTICULAR PURPOSE. See the GNU General Public License for more details.

http://www.gnu.org/licenses/

PreSTo Control Center – License page



Screen Shots

PreSTo Control Center and Database

	Α	В	С
1	R	2550	[km]
2	n_pax	99	[-]
3	m_cargo	0	[kg]
4	M_CR	0,81	[-]
5	S_LFL	1420	[m]
6	V_APP	135	[km/h]
7	S_TOFL	1463	[m]
8	n_E	2	[-]

PreSTo Control Center – **Database**

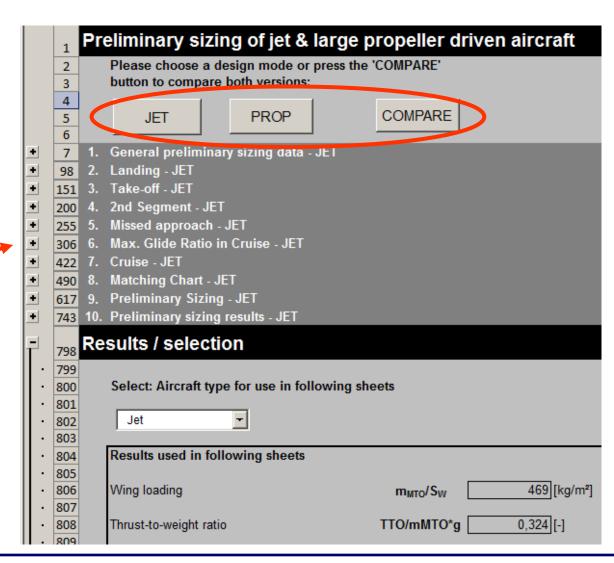




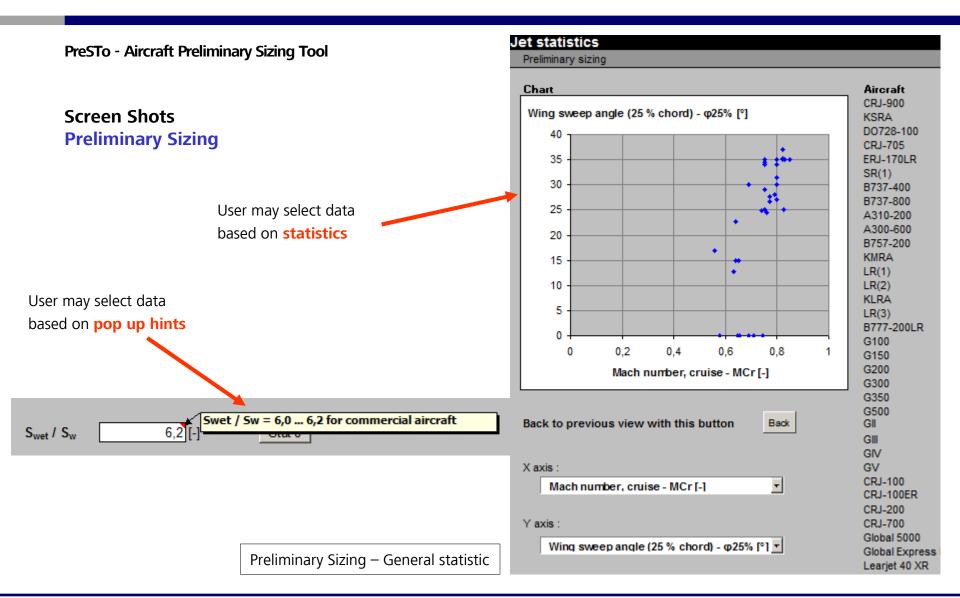
Screen Shots Preliminary Sizing

User open und close Chapters with + / - sign

Preliminary Sizing – Start page





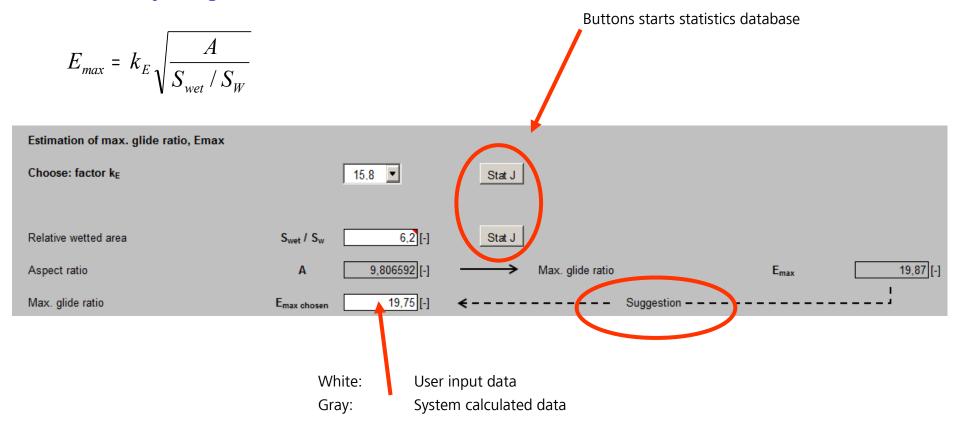




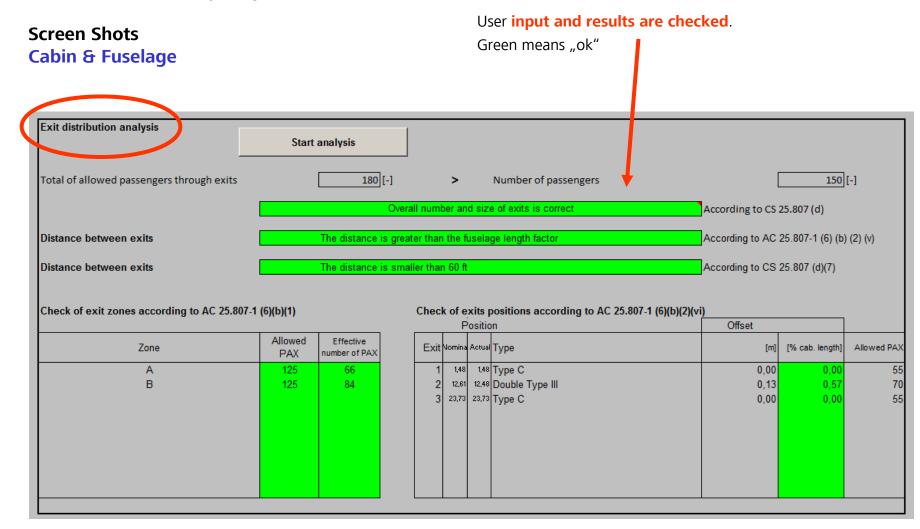


Screen Shots

Preliminary Sizing



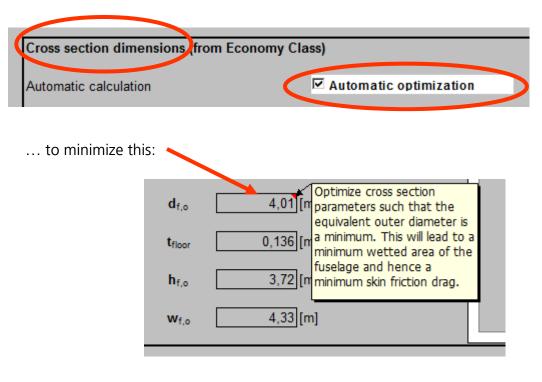


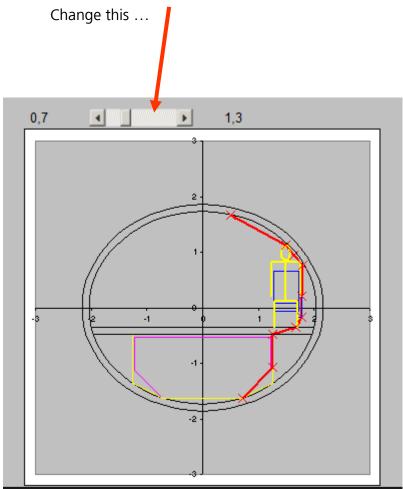




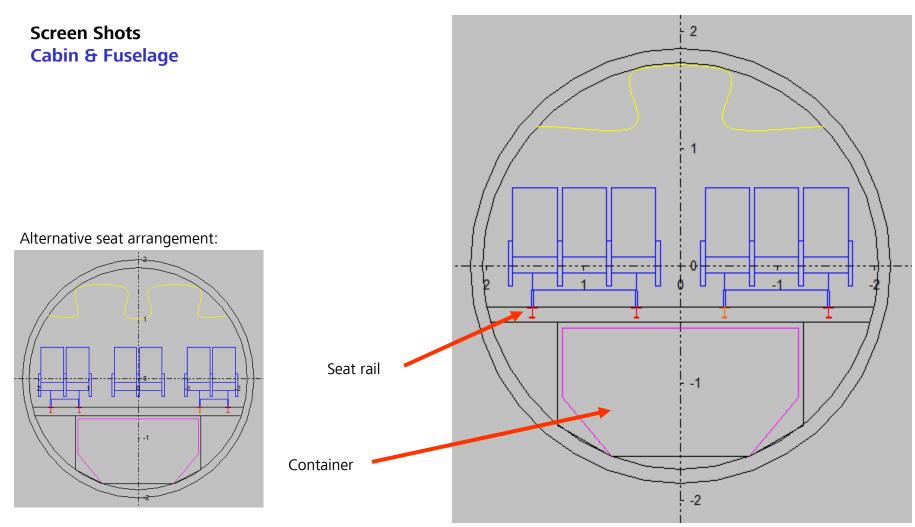
Screen Shots

Cabin & Fuselage













Screen Shots

Cabin & Fuselage



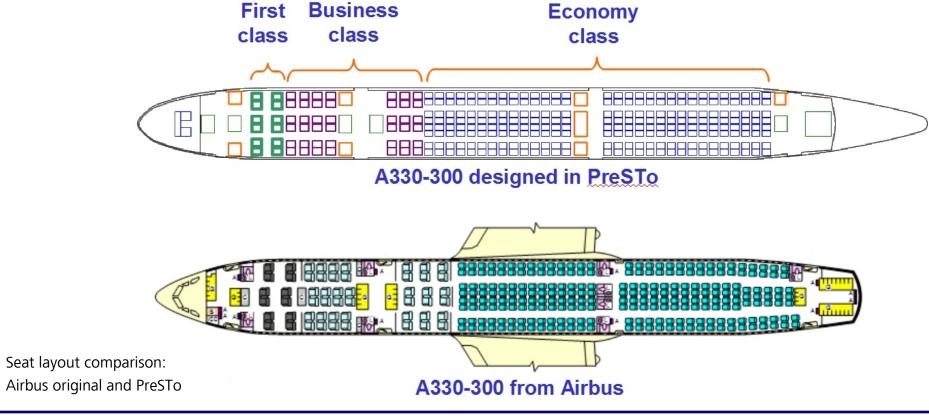
Seat layout





Screen Shots

Cabin & Fuselage

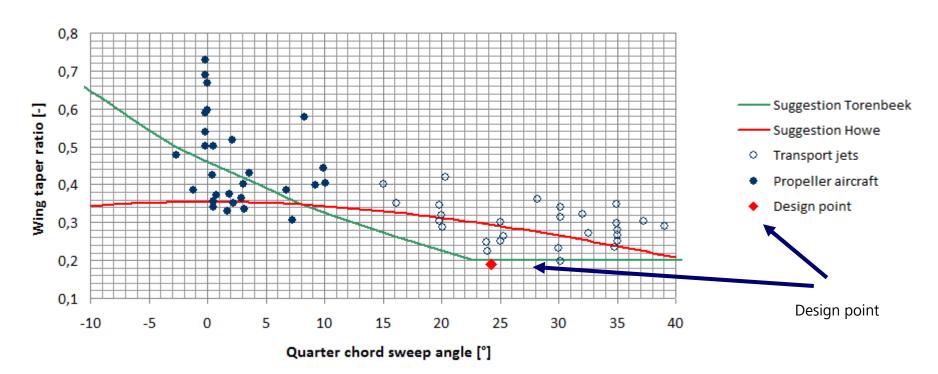






Screen Shots Wing

Taper ratio suggestion



User support with experience from industry and academia presented with respect to current design





Screen Shots

Wing



- 25 % chordline

- Wing circumference

Kinkline

Fuselage

MAC

Low Speed Ailerons

High Speed Ailerons

Preview of wing parameters





Screen Shots High Lift Final statement in high lift preliminary design

 available increase of lift coefficient due to highlift devices

 ΔC_{L,max,High_Lift}

 1,985

 required increase of lift coefficient

 ΔC_{L,max,required}

 1,757

Highlift is sufficient





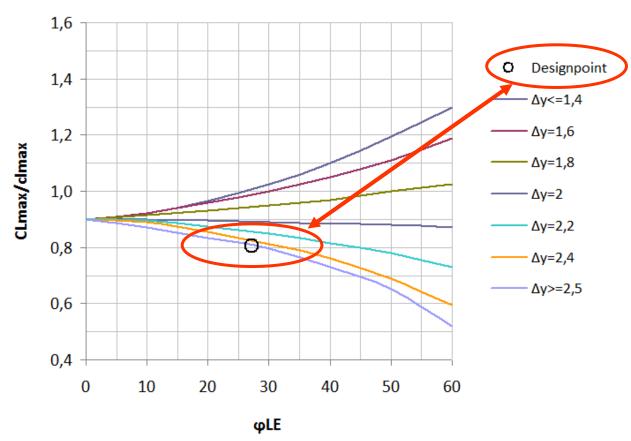


Screen Shots High Lift

Display of calculated Datcom data and **Automatic readout** of parameters

with respect of actual design point

SUBSONIC MAXIMUM LIFT OF HIGH-ASPECTED-RATIO WINGS ACCORDING TO DATCOM FIGURE 4.1.3.4-21a



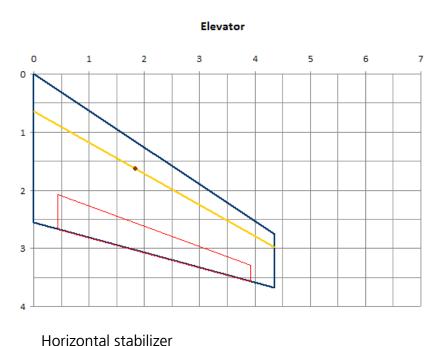




Screen Shots

Tailplane I

Preview of tail parameters



Rudder



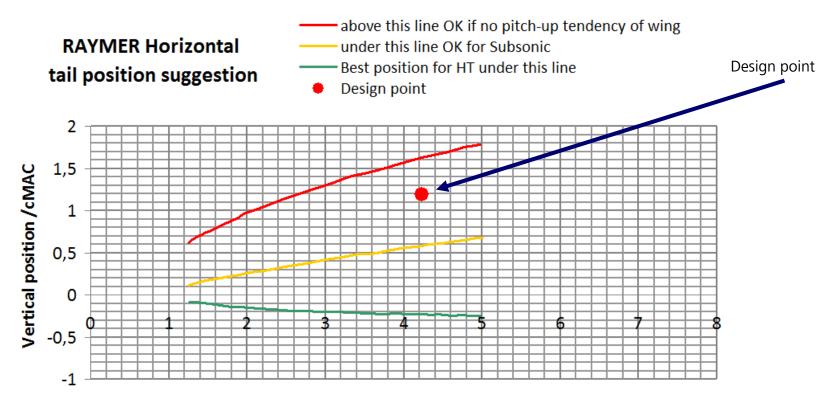
Fin



Screen Shots

Tailplane I

Showing design parameters with respect to established practise



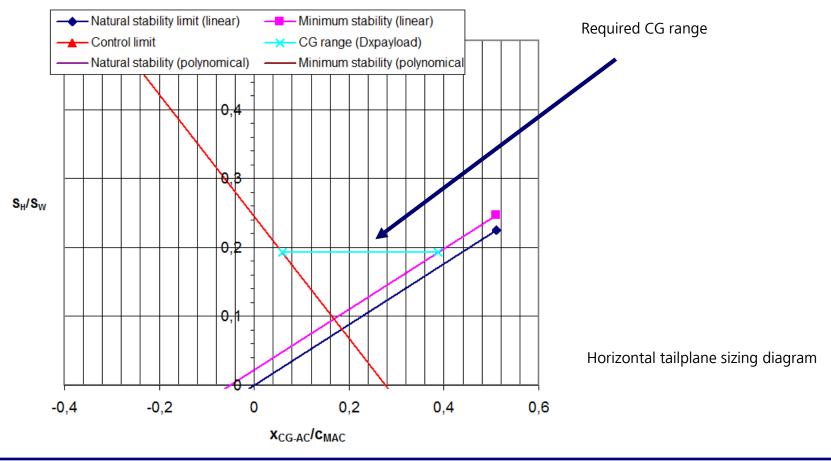
Horizontal position /cMAC





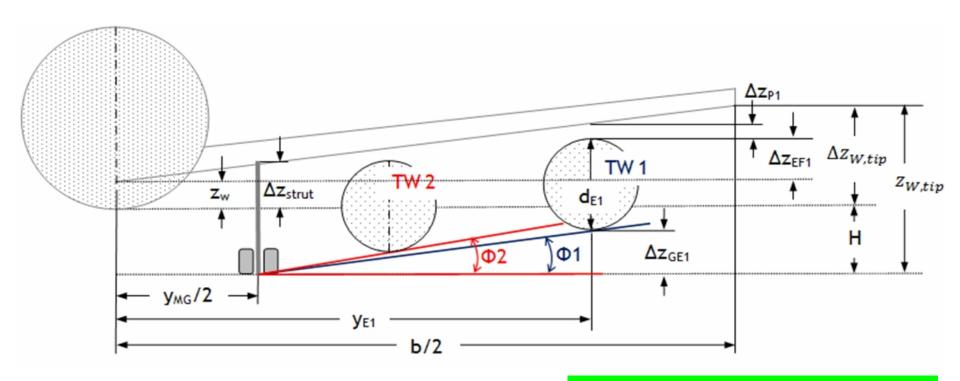
Screen Shots

Tailplane II





Screen Shots Landing Gear



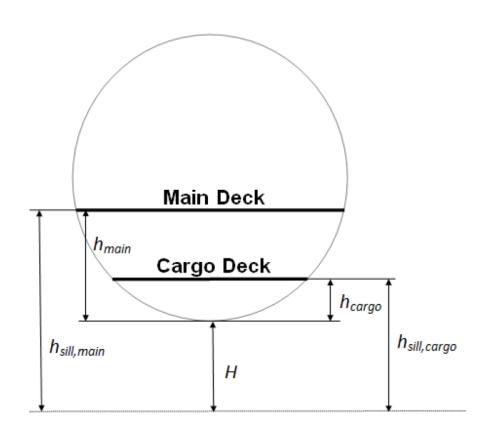
Engine ground clearance due to landing gear length

Engine 1 bank angle is OK





Screen Shots Landing Gear



Calculating sill height – an important parameter for airport compatability

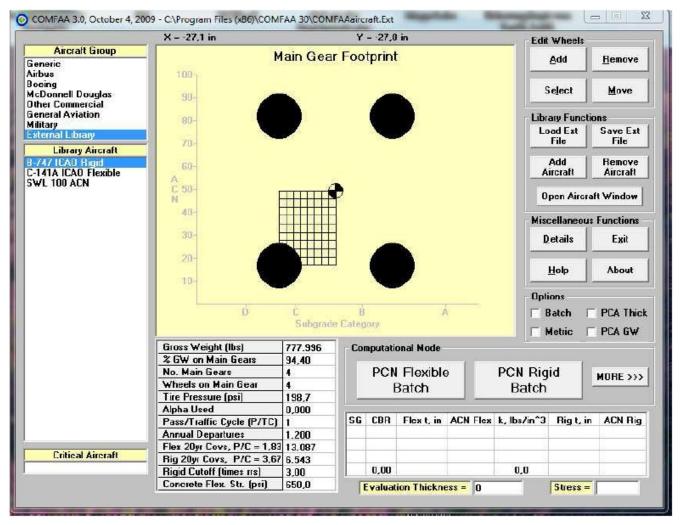




Screen Shots Landing Gear

Calculation of **ACN** values Aircraft Classification Number

COMFAA is integrated into PreSTo: o automatic input of data o COMFAA results stored in PreSTo

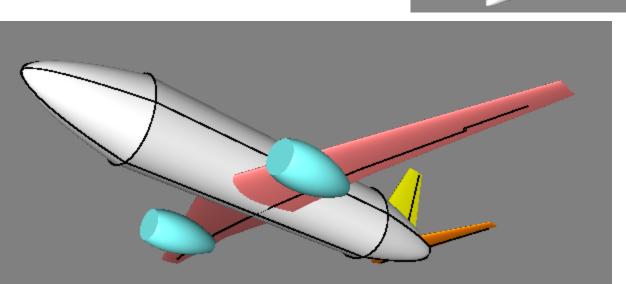






Data Export / Visualization CEASIOM



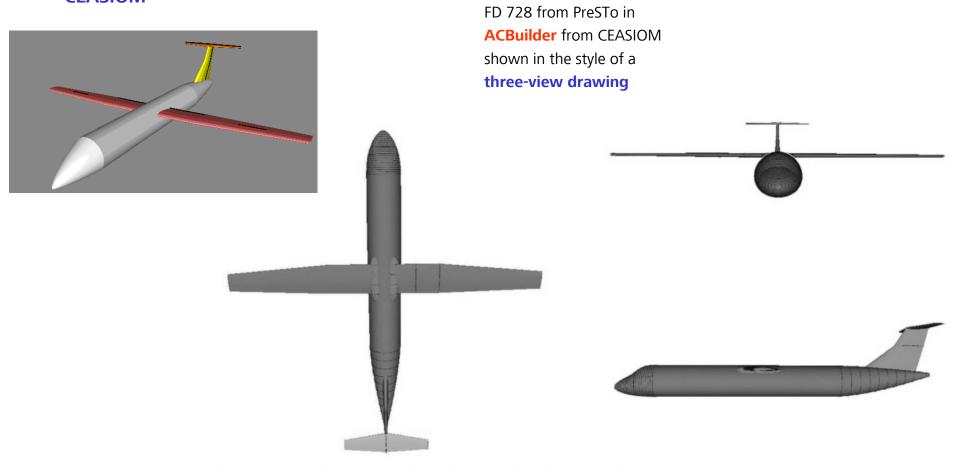








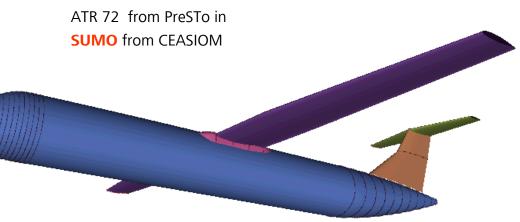
Data Export / Visualization CEASIOM

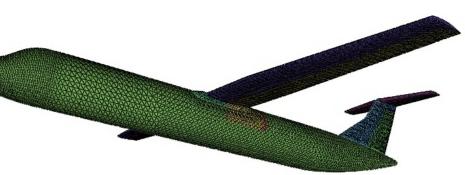




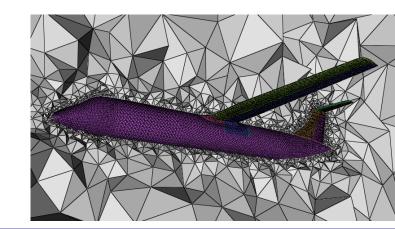


Data Export / Visualization CEASIOM





ATR 72 from PreSTo with surface and volume mesh generated by **SUMO** from CEASIOM

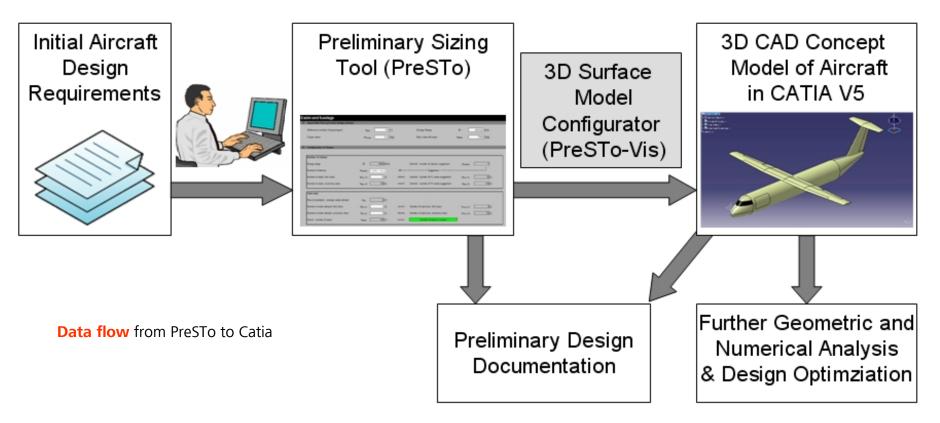






Data Export / Visualization

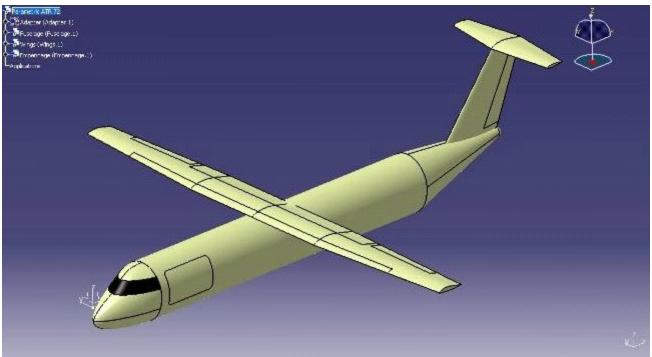
Catia





Data Export / Visualization Catia





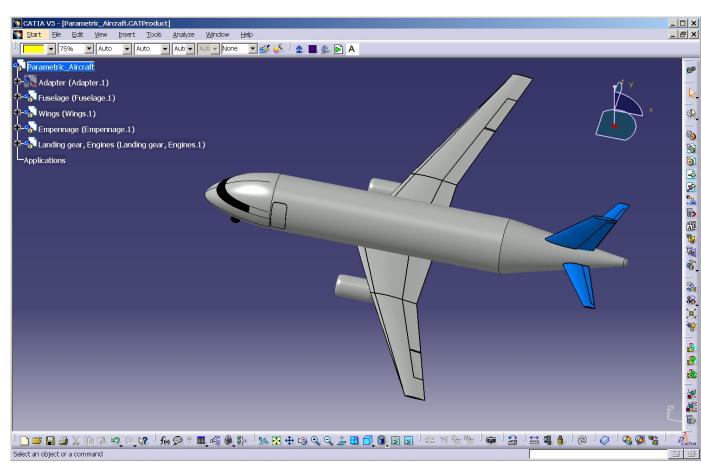
ATR 72 from PreSTo in Catia built with parametric model





Data Export / Visualization

Catia



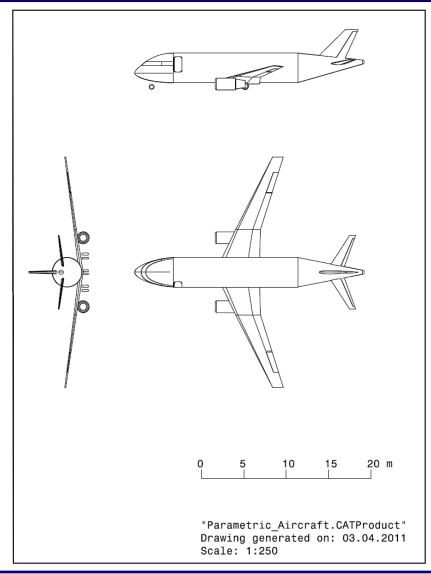
FD 728 from PreSTo in Catia built with parametric model





Data Export / Visualization Catia

FD 728 from PreSTo in Catia automatically generated three-view drawing derived from parametric model





Data Export / Visualization

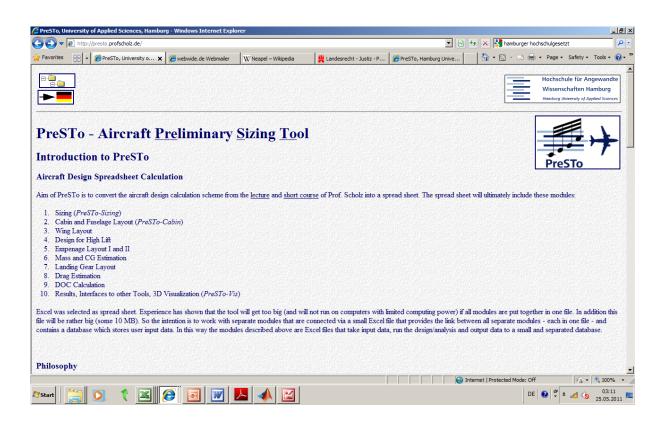
PrADO (Preliminary Aircraft Design and Optimization)







PreSTo Homepage



http://PreSTo.ProfScholz.de





PreSTo Homepage

Download PreSTo:



http://PreSTo.ProfScholz.de





Conclusions and Outlook

- PreSTo supports a very basic / standard way of aircraft design
- Interfaces are provided to higher order tools
 - CEASIOM
 - PrADO
- Visualization of the aircraft is done with outside tools:
 - CEASIOM
 - ACBuilder
 - SUMO
 - Catia
- Next steps:
 - Finish PreSTo
 - Offer for donwload: http://PreSTo.ProfScholz.de





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